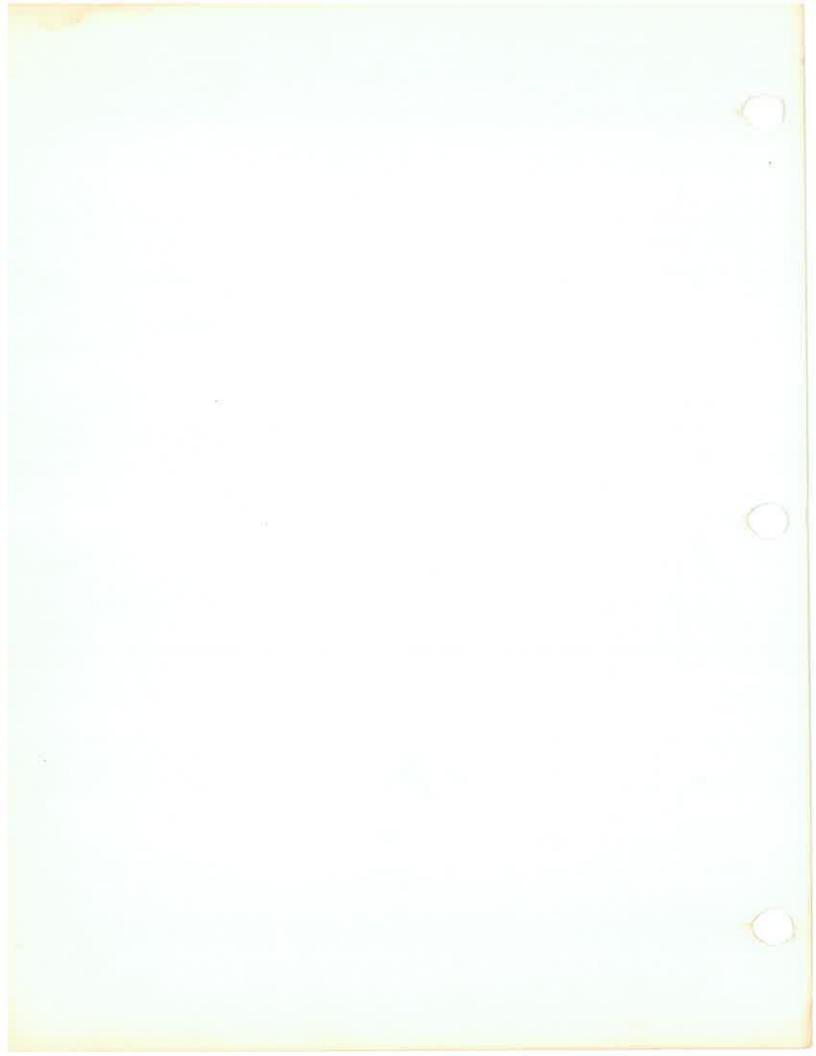
TABLE OF CONTENTS

SECTION 2

LIMITATIONS

Paragr No.	raph	Page No.
2.1	General	2-1
2.3	Airspeed Limitations	2-1
2.5	Airspeed Indicator Markings	2-2
2.7	Power Plant Limitations	2-2
2.9	Power Plant Instrument Markings	2-3
2.11	Weight Limits	2-3
2.13	Center of Gravity Limits	
2.15	Maneuver Limits	2-4
2.17	Flight Load Factors	2-4
2.19	Types of Operations	2-5
2.21	Fuel Limitations	2-5
2.23	Noise Levels	2-5
2.25	Placards	2-7



SECTION 2 LIMITATIONS

2.1 GENERAL

This section provides the "FAA Approved" operating limitations, instrument markings, color coding and basic placards necessary for operation of the airplane and its systems.

This airplane must be operated as a normal or utility category airplane in compliance with the operating limitations stated in the form of placards and markings and those given in this section and handbook.

Limitations associated with those optional systems and equipment which require handbook supplements can be found in Section 9 (Supplements).

2.3 AIRSPEED LIMITATIONS

SPEED	KIAS	KCAS
Never Exceed Speed (VNE) - Do not exceed this speed in any operation.	160	153
Maximum Structural Cruising Speed (VNO)- Do not exceed this speed except in smooth air and then only with caution.	126	122
Design Maneuvering Speed (VA) - Do not make full or abrupt control movements above this speed. At 2325 LBS. G.W.	111	108 89
At 1531 LBS. G.W.	88	

CAUTION

Maneuvering speed decreases at lighter weight as the effects of aerodynamic forces become more pronounced. Linear interpolation may be used for intermediate gross weights. Maneuvering speed should not be exceeded while operating in rough air.

Maximum Flaps Extended Speed (VFE)- Do not exceed		
this speed with the flaps extended.	103	100

ISSUED: DECEMBER 16, 1976 REVISED: NOVEMBER 20, 1981

2.5	ATRSPEED	INDICATOR I	MARKINGS
-		LUDICALUKI	

MARKING

IAS

Red Radial Line (Never Exceed)

160 KTS

Yellow Arc (Caution Range - Smooth Air Only)

126 KTS to 160 KTS

Green Arc (Normal Operating Range)

50 KTS to 126 KTS

White Arc (Flap Down)

44 KTS to 103 KTS

2.7 POWER PLANT LIMITATIONS

(a) Number of Engines

1

(b) Engine Manufacturer

Lycoming 0-320-D2A or 0-320-D3G

(c) Engine Model No.(d) Engine Operating Limits

160

(1) Maximum Horsepower(2) Maximum Rotation Speed (RPM)

2700

(3) Maximum Oil Temperature

245°F

(e) Oil Pressure

Minimum (red line)

25 PSI

Maximum (red line)

100 PSI

(f) Fuel Pressure Minimum (red line)

5 PSI

Maximum (red line)

8 PSI 100 or 100LL Aviation Grade

(g) Fuel (minimum grade)(h) Number of Propellers

1

(i) Propeller Manufacturer

Sensenich

(j) Propeller Model

74DM6-0-60 or 74DM6-0-58

(k) Propeller Diameter

Minimum

72 IN.

Maximum

74 IN.

(1) 74DM6-0-60 Propeller Tolerance (static rpm at maximum permissible

throttle setting, Sea Level, ISA)

Not above 2430 RPM Not below 2330 RPM

NOTE

Refer to the airplane maintenance manual for test procedure to determine approved static rpm under non standard conditions.

(m) 74DM6-0-58 Propeller Tolerance (static RPM at maximum permissible throttle setting, Sea Level, ISA)

Not above 2465 RPM Not below 2365 RPM

NOTE

Refer to the airplane maintenance manual for test procedure to determine approved static rpm under non standard conditions.

REPORT: VB-880

ISSUED: DECEMBER 16 1976 REVISED: SEPTEMBER 13, 1990

2.9 POWER PLANT INSTRUMENT MARKINGS

(a)	Tachometer Green Arc (Normal Operating Range) Red Line (Maximum Continuous Power)	500 to 2700 RPM 2700 RPM
(b)	Oil Temperature	750 - 24597
	Green Arc (Normal Operating Range)	75° to 245°F
	Red Line (Maximum)	243 F
(c)	Oil Pressure	(0 DGI 4- 00 DGI
	Green Arc (Normal Operating Range)	60 PSI to 90 PSI
	Yellow Arc (Caution Range) (Idle)	25 PSI to 60 PSI
	Red Line (Minimum)	25 PSI
	Red Line (Maximum)	90 PSI
(d)	Fuel Pressure	
()	Green Arc (Normal Operating Range)	.5 PSI to 8 PSI
	Red Line (Minimum)	.5 PSI
	Red Line (Maximum)	8 PSI

2.11 WEIGHT LIMITS

	NORMAL	UTILITY
(a) Maximum Weight	2325 LBS	2020 LBS
(b) Maximum Baggage	200 LBS	0 LBS

NOTE

Refer to Section 5 (Performance) for maximum weight as limited by performance.

ISSUED: DECEMBER 16, 1976 REVISED: JULY 3, 1979 REPORT: VB-880

2-3

2.13 CENTER OF GRAVITY LIMITS

(a) Normal Category

Weight Pounds	Forward Limit Inches Aft of Datum	Rearward Limit Inches Aft of Datum
2325	87.0	93.0
1950 (and less)	83.0	93.0

(b) Utility Category

Weight Pounds	Forward Limit Inches Aft of Datum	Rearward Limit Inches Aft of Datum
1950 (and less)	83.0 84.0	93.0 93.0

NOTES

Straight line variation between points given.

The datum used is 78.4 inches ahead of the wing leading edge at the inboard intersection of the straight and tapered section.

It is the responsibility of the airplane owner and the pilot to insure that the airplane is properly loaded. See Section 6 (Weight and Balance) for proper loading instructions.

2.15 MANEUVER LIMITS

(a) Normal Category - All acrobatic maneuvers including spins prohibited.

(b) Utility Category - Approved maneuvers for bank angles exceeding 60°:

	Entry Speed
Steep Turns	111 KIA S
Lazy Eights	111 KIAS
Chandelles	111 KIAS

2.17 FLIGHT LOAD FACTORS

		NORMAL	UTILITY
(a)	Positive Load Factor (Maximum)	3.8 G	4.4 G
	Negative Load Factor (Maximum)	No inverted maneuvers approved	

REPORT: VB-880

ISSUED: DECEMBER 16, 1976 REVISED: JULY 3, 1979

2.19 TYPES OF OPERATION

The airplane is approved for the following operations when equipped in accordance with FAR 91 or FAR 135.

- (a) Day V.F.R.(b) Night V.F.R.
- (c) Day I.F.R.
- (d) Night I.F.R.
- (e) Non Icing

2.21 FUEL LIMITATIONS

(a)	Total Capacity	50 U.S. GAL
(b)	Unusable Fuel	2 U.S. GAL
	The unusable fuel for this airplane has been determined	
	as 1.0 gallon in each wing in critical flight attitudes.	
(c)	Usable Fuel	48 U.S. GAL
	The usable fuel in this airplane has been determined as	
	24.0 gallons in each wing.	

2.23 NOISE LEVEL

The noise level of this aircraft is 72dBA.

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

The above statement notwithstanding, the noise level stated above has been verified by and approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, Noise Standards - Aircraft Type and Airworthiness Certification. This aircraft model is in compliance with all FAR 36 noise standards applicable to this type.

ISSUED: DECEMBER 16, 1976 REVISED: FEBRUARY 24, 1977 REPORT: VB-880

THIS PAGE INTENTIONALLY LEFT BLANK

REPORT: VB-880 2-6

ISSUED: DECEMBER 16, 1976

2.25 PLACARDS

In full view of the pilot:

"THIS AIRPLANE MUST BE OPERATED AS A NORMAL OR UTILITY CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARKINGS AND MANUALS.

ALL MARKINGS AND PLACARDS ON THIS AIRPLANE APPLY TO ITS OPERATION AS A UTILITY CATEGORY AIRPLANE. FOR NORMAL AND UTILITY CATEGORY OPERATION, REFER TO THE PILOT'S OPERATING HANDBOOK.

NO ACROBATIC MANEUVERS ARE APPROVED FOR NORMAL CATAGORY OPERATIONS. SPINS ARE PROHIBITED FOR NORMAL AND UTILITY CATEGORY."

In full view of the pilot, one of the following takeoff checklists and one of the following landing check lists will be installed:

TAKEOFF CHECK LIST

Fuel on proper tank
Electric fuel pump on
Engine gauges checked
Flaps - set
Carb heat off

Mixture set
Seat backs erect

Fasten belts/harness
Trim tab - set
Controls - free
Door - latched
Air conditioner - off

TAKEOFF CHECK LIST

Fuel on proper tank Electric fuel pump-on Engine gages checked

Flaps - set Carb. heat off Mixture set Primer locked Seat backs erect

Fasten belts/harness
Trim tab - set
Controls - free
Door - latched
Air conditioner off

LANDING CHECK LIST

Fuel on proper tank Mixture rich Electric fuel pump on

Seat backs erect

Flaps - set (103 KIAS max.)
Fasten belts/harness
Air conditioner off

LANDING CHECK LIST

Fuel on proper tank Mixture rich Electric fuel pump on

Seat backs erect.

Flaps - set (White Arc). Fasten belts harness Air conditioner off

The "AIR COND OFF" item in the above takeoff and landing check lists is mandatory for air conditioned aircraft only.

ISSUED: DECEMBER 16, 1976 REVISED: JULY 3, 1979 REPORT: VB-880

In full view of the pilot, in the area of the air conditioner control panel when the air conditioner is installed:

"WARNING — AIR CONDITIONER MUST BE OFF TO INSURE NORMAL TAKEOFF CLIMB PERFORMANCE." $\,$

Adjacent to upper door latch:

"ENGAGE LATCH BEFORE FLIGHT."

On inside of the baggage compartment door:

"BAGGAGE MAXIMUM 200 LBS"
"UTILITY CATEGORY OPERATION - NO BAGGAGE OR AFT PASSENGERS ALLOWED. NORMAL CATEGORY OPERATION
- SEE PILOT'S OPERATING HANDBOOK WEIGHT AND BALANCE SECTION FOR BAGGAGE AND AFT PASSENGER LIMITATIONS."

In full view of the pilot:

"MANEUVERING SPEED 111 KIAS AT 2325 LBS. (SEE P.O.H.)" OR "VA = 111 KIAS AT 2325 #(SEE P.O.H.)"

"UTILITY CATEGORY OPERATION - NO AFT PASSENGERS ALLOWED."

"DEMONSTRATED CROSS WIND COMPONENT - 17 KTS." or "DEMO. X-WIND 17 KTS."

In full view of the pilot when the oil cooler winterization kit is installed:

"OIL COOLER WINTERIZATION PLATE TO BE REMOVED WHEN AMBIENT TEMPERATURE EXCEEDS 50°F."

In full view of the pilot:

"UTILITY CATEGORY OPERATION ONLY."

- (1) NO AFT PASSENGERS ALLOWED.
- (2) ACROBATIC MANEUVERS ARE LIMITED TO THE FOLLOWING:

ENTRY SPEED

SPINS PROHIBITED
STEEP TURNS
LAZY EIGHTS
CHANDELLES

111 KIAS
111 KIAS

In full view of the pilot:

"WARNING — TURN OFF STROBE LIGHTS WHEN IN CLOSE PROXIMITY TO GROUND OR DURING FLIGHT THROUGH CLOUD, FOG OR HAZE."

REPORT: VB-880

ISSUED: DECEMBER 16, 1976 REVISED: JULY 3, 1979